# A DISCRETE-TIME ROBUST STATE DERIVATIVE FEEDBACK CONTROLLER FOR AN ACTIVE SUSPENSION SYSTEM

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Abstract— This work presents the design and the experimental implementation of a robust state derivative feedback (SDF) controller in an active suspension system manufactured by Quanser<sup>®</sup> (Quanser, 2009a). For this purpose, a discrete-time SDF controller is designed in the presence of parametric uncertainties by using a state derivative model within a regional pole placement approach. In the implementations, only the measurements from accelerometers are employed, a typical motivation for the use of SDF. The dynamic behavior of the active suspension system is tested for different values of the uncertain parameter within its range considered in the controller design.

Keywords— state derivative feedback, discrete-time control, polytopic uncertainty, active suspension system

**Resumo**— Este trabalho apresenta o projeto e a implementação experimental de um controlador robusto usando realimentação da derivada dos estados em um sistema de suspensão ativa fabricado por Quanser<sup>®</sup> (Quanser, 2009a). Para este propósito, um controlador em tempo discreto com realimentação derivativa é projetado na presença de incertezas paramétricas usando um modelo baseado nas derivadas dos estados em um método de alocação regional de polos. Na implementação, emprega-se apenas medidas de acelerômetros, uma motivação típica para o uso de realimentação derivativa. O comportamento dinâmico do sistema de suspensão ativa é apresentado para diferentes valores da incerteza paramétrica dentro da faixa considerada no projeto.

**Palavras-chave** realimentação da derivada dos estados, controle a tempo discreto, incerteza politópica, sistema de suspensão ativa

## 1 Introduction

Engineering applications have widely employed accelerometer as the main sensor in the instrumentation sets, due to the low operational cost, simple structure and the ease in measuring the acceleration signals, instead of displacement and velocity, in the absence of an absolute position reference (Yang et al., 1991). Examples of applications include active suspension devices (Reithmeier and Leitmann, 2003), (da Silva et al., 2013), vibration suppression of mechanical systems (Abdelaziz, 2012), driving assistance controllers (Fallah et al., 2013), vibration control of bridge cables (Duan et al., 2005), and earthquake hazard mitigation (Yang et al., 1991).

As the state variables of these systems usually correspond to displacements and velocities, the use of accelerometers has motivated a significant research in the control literature concerning state derivative feedback (SDF). In fact, by double integration of the measured accelerations, the estimated displacements may not have good accuracy, due to the propagation of errors associated to bias in the measurements (Abdelaziz, 2012) and uncertainty in the initial conditions for the integrators (Reithmeier and Leitmann, 2003). Thus, the use of SDF may be more convenient, since it employs velocities and accelerations, not displacements.

Most works concerning SDF design have been devoted to continuous-time controllers, such as pole placement in (Abdelaziz and Valášek, 2004), (Faria et al., 2009), linear quadratic regulators (LQR) in (Duan et al., 2005), (Faria et al., 2009), (Tseng and Hsieh, 2013), robust controllers formulation using linear matrix inequalities (LMIs) in (Assunção et al., 2007), (Faria et al., 2010), (da Silva et al., 2012), among others.

The discrete-time design of SDF controllers were still only developed in a few works. In (Cardim et al., 2009) and (Rossi et al., 2013), the design of SDF controllers in nominal equivalence to a given state feedback controller were presented. A direct discrete-time design of robust SDF control laws was proposed in (Rossi et al., 2018), dispensing with the need for a preliminary state feedback design. Moreover, this method allowed the design of controllers with robustness to parametric uncertainties. However, an experimental validation of this proposed technique was not exhibited in (Rossi et al., 2018), only numerical simulation results were shown.

In this context, this work presents a practical implementation of a robust discrete-time SDF controller designed by using the method proposed in (Rossi et al., 2018). This controller is applied to an active suspension system manufactured by Quanser<sup>®</sup> (Quanser, 2009a). For this purpose, a state derivative model is employed, as proposed in (Rossi et al., 2018), within a classic LMI approach for regional pole placement (Chilali and Gahinet, 1996). A robust SDF controller is then designed in the presence of parametric uncertainties (an uncertain mass of the system) and implemented using only measurements from accelerometers.

The remainder of this paper is organized as follows. Section 2 describes the discrete-time design of controllers using SDF proposed in (Rossi et al., 2018). Section 3 presents the active suspension system. The controller design is developed in Section 4 and the practical implementation and results are exhibited in Section 5. Finally, concluding remarks are shown in Section 6.

### 2 State derivative feedback in discrete-time controllers design

The method proposed in (Rossi et al., 2018) for discrete-time controller design using SDF is presented in this section.

Consider a system described by a continuoustime model of the form

$$\dot{x}(t) = \Phi_c x(t) + \Gamma_c u(t) \tag{1}$$

with the state vector  $x(t) \in \mathbb{R}^n$ , the control input  $u(t) \in \mathbb{R}^m$ , and the constant matrices  $\Phi_c \in \mathbb{R}^{n \times n}$ ,  $\Gamma_c \in \mathbb{R}^{n \times m}$ , with  $\Phi_c$  non-singular.

Assume that the system is to be controlled by using sampled measurements of the state derivative  $\dot{x}(kT), k \in \mathbb{Z}$ , where T is the sampling period. Moreover, consider that a zero order hold is employed to keep the control u(t) constant between sampling times, i.e.

$$u(t) = u(kT)^+, \ (kT)^+ \le t \le (k+1)T$$
 (2)

The superscript + in (2) is employed to indicate that the control is updated immediately after the state derivative is measured at each sampling time. Therefore, the state derivative of the system (1) at time t = kT is given by

$$\dot{x}(kT) = \Phi_c x(kT) + \Gamma_c u((k-1)T)^+$$
 (3)

Since the control is kept constant between sampling times, as in Eq. (2), the model (1) can be discretized as

$$x((k+1)T) = \Phi x(kT) + \Gamma u(kT)^+ \qquad (4)$$

with  $\Phi = e^{\Phi_c T}$  and  $\Gamma = \int_0^T e^{\Phi_c \tau} \Gamma_c \, \mathrm{d}\tau$ .

The following theorem shows that the model (4) can be reformulated in terms of the derivative of the state  $\dot{x}(kT)$  and the control input, in a suitable form for use in discrete-time control design.

**Theorem 1** Let  $\dot{x}(kT)$  denote the derivative of the state at sampling time t = kT, immediately before the control update. The discrete-time model (4) can then be recast into the following form:

$$\xi((k+1)T) = A\xi(kT) + Bu(kT)^{+}$$
 (5)

with  $\xi(kT) \in \mathbb{R}^{n+m}$ ,  $A \in \mathbb{R}^{(n+m) \times (n+m)}$  and  $B \in \mathbb{R}^{(n+m) \times m}$  defined as

$$\xi(kT) \triangleq \left[ \begin{array}{c} \dot{x}(kT) \\ u((k-1)T)^+ \end{array} \right]$$
(6)

$$A = \begin{bmatrix} \Phi & -\Phi\Gamma_c \\ 0 & 0 \end{bmatrix}, \quad B = \begin{bmatrix} \Phi\Gamma_c \\ I \end{bmatrix}$$
(7)

where 0 and I denote a matrix of zeros and an identity matrix of appropriate dimensions, respectively.

**Proof:** See in (Rossi et al., 2018). 
$$\Box$$

**Remark 1** The representation (5) derived in Theorem 1 can be employed to design control laws of the form:

$$u(kT)^{+} = F\xi(kT) \tag{8}$$

with a feedback gain matrix  $F \in \mathbb{R}^{m \times (n+m)}$ , which can be designed through standard discrete-time state-space methods. In the examples illustrated in Section 5, it is assumed that the control task starts at time k = 0 and thus the control law is initialized with  $u(-T)^+ = 0$ .

### Remark 2 (Polytopic uncertainties)

Consider a model of the form (1), with matrices  $\Phi_c$  and  $\Gamma_c$  subject to polytopic uncertainties, i.e.  $(\Phi_c, \Gamma_c) \in \Omega_{\Phi_c, \Gamma_c}$ , where  $\Omega_{\Phi_c, \Gamma_c}$  is a polytope with known vertices  $(\Phi_{c,i}, \Gamma_{c,i})$ , i = 1, 2, ..., N. In addition, let  $\Phi_i = e^{\Phi_{c,i}T}$  and assume that the sampling period T is sufficiently small so that the quadratic and higher-order terms in the power series expansion of  $e^{\Phi_{c,i}T}$  can be neglected in the uncertainty representation. Therefore, the matrices  $(\Phi, \Gamma_c)$  will lie in a polytope  $\Omega_{\Phi, \Gamma_c}$ with vertices  $(\Phi_i, \Gamma_{c,i}), i = 1, 2, \dots, N$ . Thus, in view of the product between  $\Phi$  and  $\Gamma_c$  in (7), the (A, B) matrices in (5) will belong to a polytope with  $N^2$  vertices, which are associated to the cross-products between the vertices  $\Phi_i$ , i = 1, 2, ..., N, and  $\Gamma_{c,j}$ , j = 1, 2, ..., N. In cases where T is not sufficiently small for assuming the hypothesis considered above, this issue needs to be studied in more detail. Indeed, in these cases, the use of techniques for the systematic treatment of the high order terms in the discretization procedure could be investigated, as proposed in (Braga et al., 2014). However, the first approach described herein leads to simpler design procedures and can be appropriate to meet closed-loop specifications, as will be illustrated in Section 5.

#### **3** Active suspension system

Fig. 1 shows the active suspension plant at Laboratório de Pesquisa em Controle (LPC), Faculdade de Engenharia de Ilha Solteira / Universidade Estadual Paulista "Júlio de Mesquita Filho" (FEIS/UNESP), where the experiments reported herein were carried out. It is a bench-scale model representing a classic quarter-car model controlled by an active suspension mechanism. This plant consists of three floors/plates on top of each other. The top floor (blue plate) represents the vehicle body supported above the suspension. The middle floor (red plate) corresponds to the tire. The bottom floor (silver plate) provides the road excitation in the system. A DC motor is standing between the top and middle floors to emulate an active suspension system.

A schematic model is represented in Fig. 2. The sprung mass  $M_s$  represents the mass of the vehicle body. The unsprung mass  $M_{us}$  represents the vehicle tire set. The spring  $k_s$  and the damper  $b_s$  support the body weight over the tire. The spring  $k_{us}$  and the damper  $b_{us}$  model the stiffness of the tire in contact with the road. The force  $F_c$  controls the motor (actuator) connected between the masses  $M_s$  and  $M_{us}$ , which represents the active suspension mechanism used to reduce the vibrations caused by chances on the bottom floor  $z_r$ . The control actuator  $F_c$  is limited to work in the range -39.5 N  $\leq F_c \leq 39.5$  N.

The system dynamics can be described by a continuous-time state equation of the form (Quanser, 2009a):

$$\dot{x}(t) = \Phi_c x(t) + \Gamma_c u(t) + \Gamma_{c,d} u_d(t) \qquad (9)$$

with

$$x(t) = \begin{bmatrix} z_s(t) - z_{us}(t) \\ \dot{z}_s(t) \\ z_{us}(t) - z_r(t) \\ \dot{z}_{us}(t) \end{bmatrix}$$
(10)

$$u(t) = F_c, \quad u_d(t) = \dot{z}_r(t) \tag{11}$$

$$\Phi_{c} = \begin{bmatrix} 0 & 1 & 0 & -1 \\ -\frac{k_{s}}{M_{s}} & -\frac{b_{s}}{M_{s}} & 0 & \frac{b_{s}}{M_{s}} \\ 0 & 0 & 0 & 1 \\ \frac{k_{s}}{M_{us}} & \frac{b_{s}}{M_{us}} & -\frac{k_{us}}{M_{us}} & -\frac{b_{s}+b_{us}}{M_{us}} \end{bmatrix}$$
(12)

$$\Gamma_{c} = \begin{bmatrix} 0\\ \frac{1}{M_{s}}\\ 0\\ -\frac{1}{M_{us}} \end{bmatrix}, \quad \Gamma_{c,d} = \begin{bmatrix} 0\\ 0\\ -1\\ \frac{b_{us}}{M_{us}} \end{bmatrix}$$
(13)

where  $z_s$  and  $z_{us}$  denote the vertical displacements of masses  $M_s$  and  $M_{us}$ , respectively, and  $\dot{z}_s$  and  $\dot{z}_{us}$  represent the corresponding velocities. The first state represents the suspension deflection. The third state represents the tire deflection. The control input u is composed by the force  $F_c$ . The input  $u_d$  is the bottom floor excitation represented by its velocity  $\dot{z}_r$ . The system parameters are  $M_s = 2.45$  kg,  $M_{us} = 1$  kg,  $k_s = 900$  N/m,  $k_{us} = 2500$  N/m,  $b_s = 7.5$  Ns/m and  $b_{us} = 5$  Ns/m.



Figure 1: Active suspension system at LPC, FEIS/UNESP.



Figure 2: Schematic model of the active suspension system represented by double mass-springdamper.

In the system, there is a payload mass (brass) removable in the vehicle body mass  $M_s$ . It consists of two identical weight units, each one weighting 0.4975 kg. The  $M_s$  value corresponds to the total mass ( $M_s = 2.45$  kg), which includes the total payload mass. Without the payload mass,  $M_s = 1.455$  kg. Therefore, the  $M_s$  mass may be uncertain in a range of 1.455 kg  $\leq M_s \leq 2.45$  kg (without or with the two weight units).

This active suspension system has two accelerometers measuring the accelerations  $\ddot{z}_s(kT)$ and  $\ddot{z}_{us}(kT)$  of the masses  $M_s$  and  $M_{us}$ , respectively. The velocities  $\dot{z}_s(kT)$  and  $\dot{z}_{us}(kT)$  can be estimated by integration of the respective acceleration signals. Moreover, the motion of the bottom plate  $(z_r)$  and the middle plate  $(z_{us})$  is tracked by two encoders. A third encoder measures the motion of the top plate relative to the middle one  $(z_s - z_{us})$ . However, these signals from the encoders will not be used in the control system, they will only be used to register the results.

It is worth mentioning that the original active suspension system from Quanser<sup>®</sup> does not have an accelerometer to measure  $\ddot{z}_{us}(t)$ . For implementation of SDF, the addition of this accelerometer to the system was requested by the researchers from LPC, for the manufacturer.

### 4 Controller design

In order to decrease the oscillations caused by changes on the bottom floor  $z_r$ , a regulator control system was designed for the active suspension system. Since the acceleration signals of the system were measured and the mass  $M_s$  was subject to uncertainties, an SDF control law in the presence of parametric uncertainties was developed. For this purpose, the state derivative representation (5) proposed in Theorem 1 was employed within the classic LMI approach for regional pole placement presented in (Chilali and Gahinet, 1996) (see Appendix A).

In the experimental tests that will be performed,  $z_r$  is piecewise constant. In the time interval where  $z_r$  is constant,  $\dot{z}_r$  is zero, while during the changes on the bottom floor  $z_r$  from a constant value to another,  $\dot{z}_r$  is different from zero. However, a change occurs in a small time interval. Therefore, the motion of  $z_r$  is considered fast enough so that the transient response of the signal  $z_r$  can be disregarded in the analysis. Then, in each part where the control regulation will be carried out,  $\dot{z}_r$  is considered zero.

By considering  $\dot{z}_r = 0$ , the model (9) can be described by a state equation of the form (1), with x(kT),  $\Phi_c$ ,  $\Gamma_c$  as in (10), (12), (13) respectively. As the  $M_s$  mass is uncertain in a range of 1.455 kg  $\leq M_s \leq 2.45$  kg, the model is of the form (1) with  $(\Phi_c, \Gamma_c) \in Co\{(\Phi_{c,1}, \Gamma_{c,1}), (\Phi_{c,2}, \Gamma_{c,2})\}$ , where  $M_s = 1.455$  kg (without payload mass) for  $(\Phi_{c,1}, \Gamma_{c,1})$  and  $M_s = 2.45$  kg (with total payload mass) for  $(\Phi_{c,2}, \Gamma_{c,2})$ .

Table 1 presents the eigenvalues of matrices  $\Phi_{c,1}$  and  $\Phi_{c,2}$ , as well as the corresponding damping ratios ( $\zeta$ ), natural frequencies ( $\omega_n$ ) and natural oscillation periods  $T_n = 2\pi/\omega_n$ . As can be observed, the plant has two 2nd-order modes, with dynamics features that depend on the uncertain mass  $M_s$ . For discrete-time control purposes, it was employed the default sampling period adopted in the software package provided by Quanser<sup>®</sup>, T = 1 ms, which is 100 times smaller than the smallest  $T_n$  value, as shown in Table 1.

Table 1: Eigenvalues of the continuous-time model vertices, with corresponding damping ratios ( $\zeta$ ), natural frequencies ( $\omega_n$ ) and natural oscillation periods  $T_n = 2\pi/\omega_n$ .

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	Eigenvalues	ζ	$\omega_n(rad/s)$	$T_n(s)$
$\Phi_{c,1}$	$-7.50 \pm j59.3$	0.12	59.8	0.10
	$-1.33 \pm j20.8$	0.06	20.8	0.30
Ф	$-6.95\pm j58.7$	0.12	59.1	0.11
$\Psi_{c,2}$	$-0.83\pm j16.2$	0.05	16.2	0.40

In light of Theorem 1, the resulting discretetime plant model can be cast into the form (5), with A, B as in (7) and

$$\xi(kT) = \begin{bmatrix} \dot{z}_{s}(kT) - \dot{z}_{us}(kT) \\ \ddot{z}_{s}(kT) \\ \dot{z}_{us}(kT) \\ \ddot{z}_{us}(kT) \\ u((k-1)T)^{+} \end{bmatrix}$$
(14)



Figure 3: (a) Open-loop poles of the state derivative model for the active suspension system. The dotted lines correspond to curves of constant damping ratio  $\zeta$ . (b) Closed-loop poles, with the boundary of the allocation region indicated as a thick line. The open-loop and closed-loop poles in the proximity of the unit circle are shown as inset.

Fig. 3a presents the open-loop poles of the discrete-time model (i.e. the eigenvalues of A), ob-

tained by varying the parameter  $M_s$  in the range  $1.455 \leq M_s \leq 2.45$  kg. It is worth noting that there is a pole at the origin for each case, which is associated to the row of zeros for the past control input in the structure of A defined in (7).

The control problem considered herein consists of designing a control law of the form (8), so that the closed-loop poles are placed in a desired region for any value of the uncertain parameter  $M_s$ . For stability in discrete-time, the conventional desired location of the eigenvalues is inside the unit circle in the complex plane. However, for a better transient behavior of the system, a more restricted region inside the unit circle may be selected for pole placement, in order to include performance constraints. For this purpose, in this work, the allocation region was chosen as a circle of radius r = 0.395 centered at (0.6, 0) (depicted as a thick line in Fig. 3b). More details about the choice of the allocation region can be found at (Rossi, 2018).

The gain matrix F was then obtained by using the LMI approach of (Chilali and Gahinet, 1996) described in Appendix A, with  $\xi(kT)$ , A, B in place of x(kT),  $\Phi$ ,  $\Gamma$ , respectively. As can be seen in the plant dynamics described in (12)-(13), the uncertainty in the parameter  $M_s$  affects both  $\Phi_c$ and  $\Gamma_c$ . Therefore, as discussed in Remark 2, the (A, B) matrices belong to a polytope with  $N^2 = 4$ vertices formed from the pairwise combinations of  $\Phi_1$ ,  $\Phi_2$  and  $\Gamma_{c,1}$ ,  $\Gamma_{c,2}$ . A feasible solution to the LMIs involved in the pole placement problem was obtained by using the Robust Control Toolbox<sup>TM</sup> function "feasp", resulting in the following gain F:

$$F = \begin{bmatrix} -15.296 & -0.109 & -10.331 & -0.011 & 0.779 \end{bmatrix}$$
(15)

Fig. 3b shows the closed-loop poles (eigenvalues of A + BF), again obtained by varying the parameters  $M_s$  in the range of  $1.455 \leq M_s \leq 2.45$  kg. An extended view of the open-loop and closed-loop poles in the proximity of the unit circle are shown as an inset, for a better visualization. A comparison with Fig. 3a reveals that the closed-loop poles are indeed with larger damping ratios  $\zeta$ , aiming at a suppression of the oscillations.

#### 5 Practical Implementations and Results

For experimental implementations in the active suspension system, the Matlab<sup>®</sup>/Simulink<sup>®</sup> Software is connected to Quanser<sup>®</sup>'s QUARC<sup>®</sup> Real-Time Control Software, which enables the realtime control application directly from Simulinkdesigned controllers. The velocities and accelerations signals were used for feedback. The acceleration signals were measured by accelerometers and filtered, in order to remove bias and high frequency noises, employing filters adopted in the software package provided by Quanser<sup>®</sup>. The velocities were estimated by using suitable integrating filters developed at LPC.

As an excitation signal,  $z_r(t)$  was adopted to produce a square wave signal, with amplitude 0.02m, frequency of 1/3 Hz with pulse width of 50%, for the introduction of changes on the bottom floor (disturbances), as discussed in (Quanser, 2009a). By using the designed robust SDF controller, three cases were investigated, each one with a different mass  $M_s$  coupled in the active suspension system.



Figure 4: Open-loop and closed-loop responses for the system with  $M_s = 2.45$  kg.



Figure 5: Open-loop and closed-loop responses of the accelerations for the active suspension system with total payload mass ( $M_s = 2.45$  kg).

First, the two weight units were coupled in the active suspension system ( $M_s = 2.45$  kg). Fig. 4 presents the resulting dynamic behavior of the system. A vertical dotted line indicates the time t = 10s when the control starts to operate (closed-loop). As can be seen, even in open-loop, the system is stable. However, without the control action, the displacements  $z_s$  and  $z_{us}$  of the masses  $M_s$  and  $M_{us}$  present large oscillations. By using the robust SDF controller, these oscillations were significantly reduced, as shown in the closed-loop responses, with the overshoot and settling time attenuated. Fig. 5 shows the filtered acceleration signals  $\ddot{z}_s$  and  $\ddot{z}_{us}$ . The control signal effort for the active suspension system with total payload mass ( $M_s = 2.45$  kg) is illustrated in Fig. 6.



Figure 6: Control input for the system with  $M_s = 2.45$  kg.

It is worth mentioning that the presence of offset in the displacement responses is due to nonlinearities that occur in the actual active suspension system, such as dry friction, which is not considered in the design model.

After removing a weight unit, the next test was implemented, for the system with half of the payload mass ( $M_s = 1.9525$  kg). Fig. 7 and Fig. 8 show the displacements and accelerations of the system in this condition, respectively. Fig. 9 presents the corresponding control signal effort.



Figure 7: Open-loop and closed-loop responses for the system with  $M_s = 1.9525$  kg.



Figure 8: Open-loop and closed-loop responses of the accelerations for the active suspension system with half of the payload mass ( $M_s = 1.9525$  kg).



Figure 9: Control input for the system with  $M_s = 1.9525$  kg.

Lastly, for the third implementation, the other weight unit was also removed, resulting in no payload mass coupled to the active suspension system ( $M_s = 1.455$  kg). The dynamic behavior of the system without the payload mass is exhibited in Fig. 10 and Fig. 11. The corresponding control signal effort is shown in Fig. 12.



Figure 10: Open-loop and closed-loop responses for the system with  $M_s = 1.455$  kg.



Figure 11: Open-loop and closed-loop responses of the accelerations for the active suspension system without payload mass ( $M_s = 1.455$  kg).



Figure 12: Control input for the system with  $M_s = 1.455$  kg.

In this quarter car model, the acceleration  $\ddot{z}_s$  of the sprung mass  $M_s$  can be a measure for the ride comfort, which is related to vehicle body motion sensed by the passengers (Quanser, 2009b). An analysis of the acceleration  $\ddot{z}_s$  obtained for the system in open and closed loop, for different values of  $M_s$ , is shown in Table 2. As can be seen, in closed-loop, the acceleration was decreased compared to the open-loop response, which would result in a more comfortable ride for the passengers in an actual vehicle.

Table 2: Maximum absolute value and root mean square (RMS) value of the acceleration  $\ddot{z}_s$  for the open and closed-loop system with different values of  $M_s$ .

$M_s$	Loop status	$max  \ddot{z}_s $	$RMS[\ddot{z}_s(t)]$
(kg)		$(m/s^2)$	$(m/s^2)$
9.45	Open	3.79	1.54
2.40	Closed	2.84	0.75
1.0525	Open	4.49	1.77
1.9020	Closed	3.26	0.84
1 455	Open	5.57	2.05
1.400	Closed	3.63	0.93

Thus, the designed SDF controller was able to improve the dynamic behavior of the active suspension system with robustness to the uncertainty in the mass  $M_s$ . For different values of  $M_s$ , within its range considered in the controller design, the closed-loop responses presented significant reduction of the oscillations caused by changes on the bottom floor  $z_r$ .

#### 6 Conclusion

This paper presented a practical implementation of the robust discrete-time SDF control law proposed in (Rossi et al., 2018) in an active suspension system. By considering the measurements only from accelerometers and the presence of an uncertain mass of the vehicle body, an SDF controller was designed with robustness to parametric uncertainties using a state derivative design model. The control problem consisted of reducing the oscillations caused by changes on the bottom floor. The results exhibited that the designed SDF controller improved the dynamic behavior of the active suspension system, for different values of the uncertain mass. The use of filters to remove bias and high frequency noises from the signals measured by the accelerometers did not compromise the results. However, future investigations could be carried out to obtain explicit robustness with respect to the presence of the filters. Moreover, future implementations could be concerned with the active suspension system subject to input time delay or fault in the actuator.

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# A Regional Pole Placement (CHILALI; GAHINET, 1996)

Consider a discrete-time model of the form

$$x((k+1)T) = \Phi x(kT) + \Gamma u(kT), \qquad (16)$$

where  $x(kT) \in \mathbb{R}^n$ ,  $u(kT) \in \mathbb{R}^m$  are the state and input vectors at time kT. Matrices  $(\Phi, \Gamma)$  are assumed to belong to a polytope  $\Omega$  with known vertices  $(\Phi_i, \Gamma_i), i = 1, 2, ..., N$ .

Moreover, let  $\mathcal{D}$  be a region in the complex plane described by

$$\mathcal{D} = \{ z \in \mathbb{C} \mid \alpha + z\beta + \bar{z}\beta^T < 0 \}$$
(17)

where  $\bar{z}$  denotes the complex conjugate of z and  $\alpha$ ,  $\beta$  are  $(p \times p)$  matrices of real-valued coefficients, with  $\alpha$  symmetrical.

If there exist matrices  $X = X^T \in \mathbb{R}^{n \times n}$  and  $L \in \mathbb{R}^{m \times n}$  such that the following LMIs are satisfied (Chilali and Gahinet, 1996):

$$\alpha \otimes X + \beta \otimes (\Phi_i X + \Gamma_i L) + + \beta^T \otimes (\Phi_i X + \Gamma_i L)^T < 0, \ i = 1, \dots, N$$
(18)  
$$X > 0$$
(19)

then a control law of the form u(kT) = Fx(kT), with  $F = LX^{-1}$ , will place the closed loop poles inside  $\mathcal{D}$ , for any  $(\Phi, \Gamma) \in \Omega$ . The symbol  $\otimes$  denotes the Kronecker product of matrices.

A particular case consists of placing the closed loop poles inside a circle of radius r and center  $(\chi_0, 0)$ , i.e.  $\mathcal{D} = \{z = (\chi + j\nu) \mid (\chi - \chi_0)^2 + \nu^2 < r^2\}$ . For this purpose, Schur's complement can be used to rewrite the inequality  $(\chi - \chi_0)^2 + \nu^2 < r^2$  as

$$\begin{bmatrix} -r & -\chi_0 + z \\ -\chi_0 + \bar{z} & -r \end{bmatrix} < 0$$
 (20)

which can be cast into the form  $\alpha + z\beta + \bar{z}\beta^T < 0$ of (17), with  $\alpha$  and  $\beta$  given by (Rossi et al., 2018):

$$\alpha = \begin{bmatrix} -r & -\chi_0 \\ -\chi_0 & -r \end{bmatrix}, \quad \beta = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix}$$
(21)